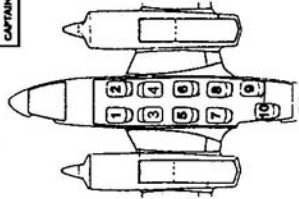


VH-ZUM

PA31-350
LOAD SYSTEM DMA-354 ISSUE 3



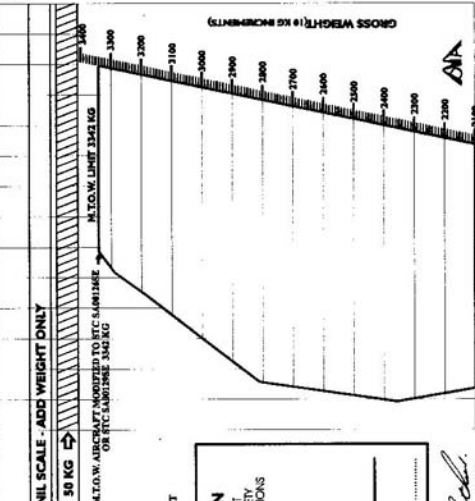
FLIGHT No.	DATE
FROM	TO
CAPTAIN	



AIRCRAFT BASE WEIGHT	WEIGHT kg	INDEX
SEATS 1 & 2 (CREW)	85 KG	↔
SEATS 3 & 4	170 KG	↔
SEATS 5 & 6	85 KG	↔
SEATS 7 & 8	85 KG	↔
SEAT 9	85 KG	↔
SEAT 10	85 KG	↔
NOSE	20 KG	↔
AFT CABIN	20 KG	↔
WING LOCKERS (85KG MAX EACH SIDE)	20 KG	↔

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MAX WT. AVAIL. ABZ	50 KG
REF. CAO Section 30.7	
TAKE-OFF WT.	
LANDING WT.	



CERTIFICATION
I AM ADVISED THAT THE AIRCRAFT IS CORRECTLY LOADED FOR SAFETY IN ACCORDANCE WITH REGULATIONS SPECIFIED BY CASA.

FLIGHT MANUAL COPY MUST BE COLOURED ORIGINAL.

DATE..... TIME.....

CAPTAIN/CO-PILOT.....

METHOD OF USE (SEE 6.2) ON RNH MARGIN OF SHEET)

- ORIGIN THE SAME INDEX IS SHOWN FROM THIS POINT ON THE INDEX SCALE. THE INDEX MUST BE READ FROM THE BOTTOM LINE, ALONG THE SCALE. THE INDEX MUST BE READ FROM THE TOP TO THE OTHER SCALES IN THE SAME MANNER. EACH SCALE IN THE SAME MANNER.
- DRAW A VERTICAL LINE FROM THE LAST COMBINATION SCALE IN USE DOWN TO THE C/G ENVELOPE INCLUDING THE FUEL/AIR FUELING. REPORT THE POSITION FROM THE C/G ENVELOPE INCLUDING THE FUEL/AIR FUELING AT APPROXIMATE 1% OF THE SCALE.
- AND DRAW AN APPROXIMATE ALLOWANCE FOR FUEL BEFORE DROPPING ANOTHER VERTICAL LINE TO THE C/G ENVELOPE, STOPPING AT THE TWE-OFF WEIGHT.

NOTE:
THIS SHEET INCORPORATES A NON-STANDARD DATA.
DO NOT USE DATA RELATING TO THE STANDARD DATA.
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P. J. M. Smith
AQ45

WYA-3231/1000 PA200

